

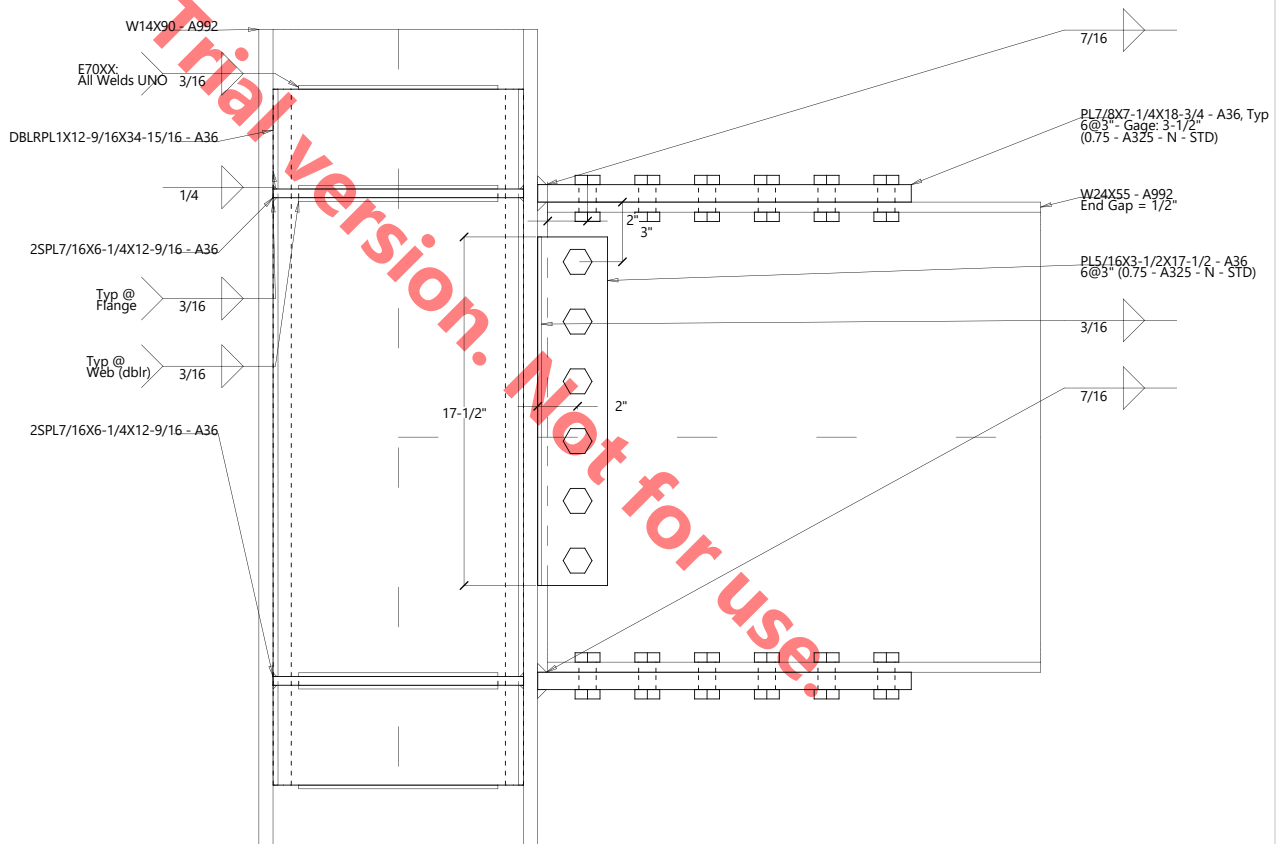
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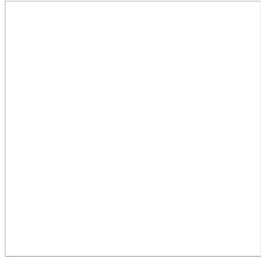
Sample Configuration  
  
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**Front View**





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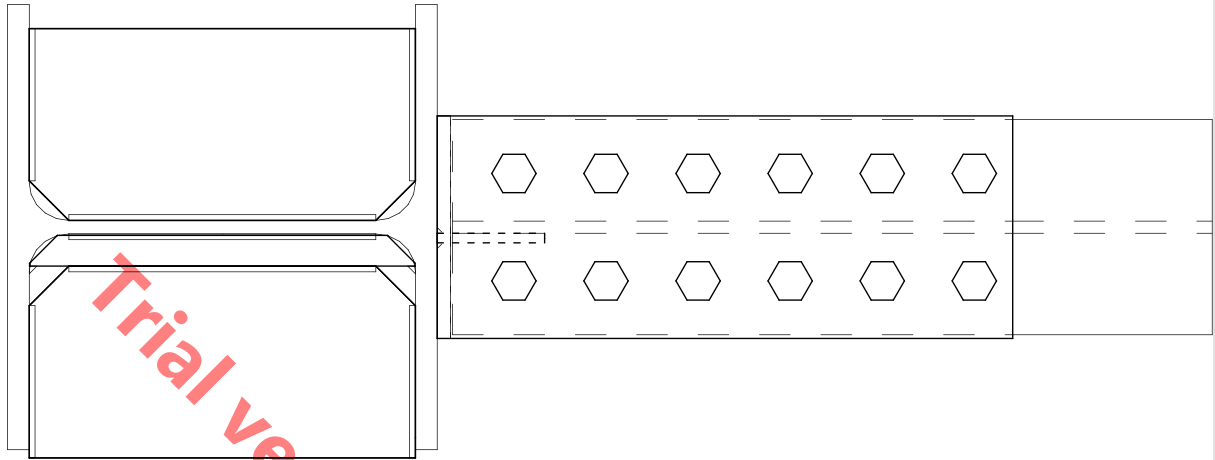
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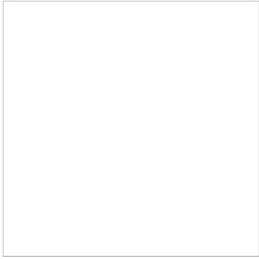
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Top View



**Trial version. Not for use.**



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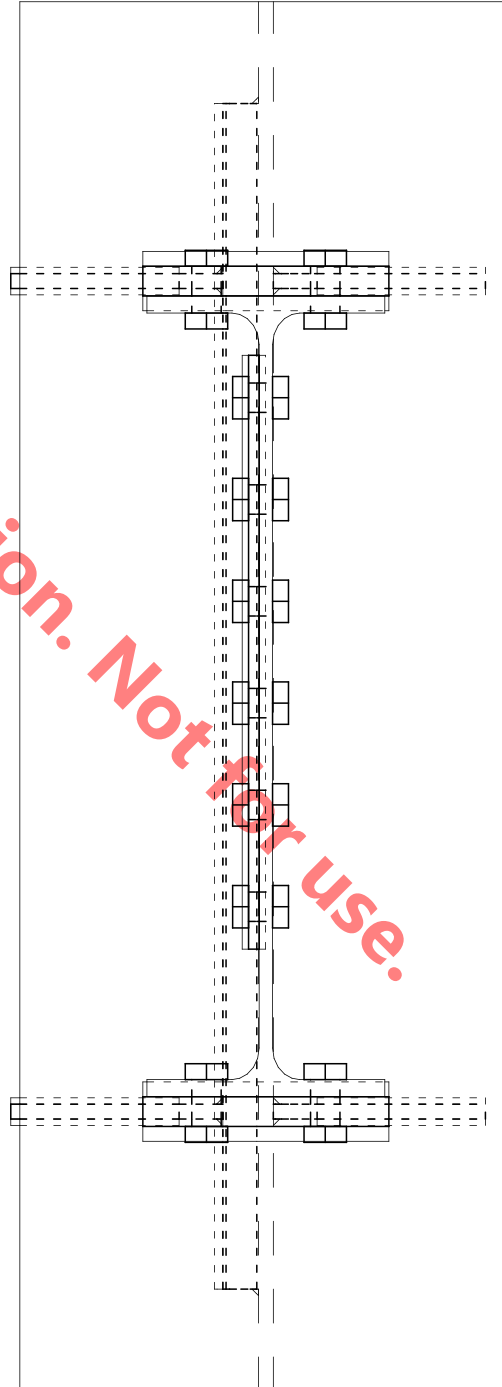
Sample Configuration

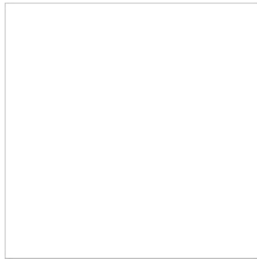
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Right Side View





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**BASIC DETAILS OVERVIEW**

**Joint Configuration:** Beam to Column Flange

**Member:** Column  
**Section:** W14X90  
**Material:** A992

**Member:** Right Side Beam  
**Section:** W24X55  
**Material:** A992

**DETAILED CALCULATION REPORT**

Beam Connection to Column Flange  
Column: W14X90 - A992

Right Side Beam: W24X55 - A992  
Moment: 250 k-ft.  
Shear: 60 kips  
Axial Force (Hc): 0 kips  
Axial Force (Ht): 0 kips

All Welds Are E70XX

**RIGHT SIDE BEAM**

**1. RIGHT SIDE BEAM - W24X55 MOMENT CONNECTION**

**1.a. Moment Connection Using Flange Plate:**

Flange Force (Ff):  
 $= P / 2 + M / d$   
 $= 0 / 2 + 3000 / 23.6$   
 $= 127.1186$  kips  
Top Plate: 18.75 in. X 7.25 in. X 0.875 in.  
Bottom Plate: 18.75 in. X 7.25 in. X 0.875 in.  
Plate Material: A36  
Bolts on Flange: 12 Bolts - (0.75 - A325 - N - STD) in 2 Lines  
Bolt Holes on Plate: 0.8125 in. Lateral X 0.8125 in. Longitudinal  
Bolt Holes on Flange: 0.8125 in. Lateral X 0.8125 in. Longitudinal

**1.b. Check Beam:**

Beam Flange Effective Area:  
 $A_{fg} = t_f * b_f = 0.505 * 7.01 = 3.540$  in<sup>2</sup>  
 $A_{fn} = t_f * (b_f - N_t * (d_h + 0.0625)) = 0.505 * (7.01 - (2 * (0.8125 + 0.0625))) = 2.6563$  in<sup>2</sup>  
 $F_y / F_u \leq 0.8$  ----  $Y_t = 1$   
 $F_u * A_{fn} = 65 * 2.6563 = 172.6595$  kips  
 $Y_t * F_y * A_{fg} = 1 * 50 * 3.540 = 177.0025$  kips  
 $M_n = F_u * A_{fn} * S_x / A_{fg} = 65 * 2.6563 * 114 / 3.540$

$= 5560.1426$  kips/in.  
 **$(1 / FS) M_n = (1 / 1.67) * M_n = 277.4522 \geq 250$  k-ft. (OK)**

**1.c. Check Bolts:**

**Spacing (s) = 3 ≥ Minimum Spacing = 2 in. (OK)**  
Edge Distance on Plate Parallel to Beam Axis (el):  
 $= 1.25 \geq 1.25$  in. (OK)  
Edge Distance on Plate Transverse to Beam (et):  
 $= 1.875 \geq 1.25$  in. (OK)  
Edge Distance on Beam Parallel to Beam Axis (el):  
 $= 2 \geq 1.25$  in. (OK)  
Edge Distance Transverse to Beam (et):  
 $= 1.755 \geq 1$  in. (OK)

**1.d. Allowable Bolt Shear Strength - Flange Plate**

$(1 / FS) R_n = n N * n_{SP} * n_B * F_v$   
 $= 1 * 1 * 12 * 10.6028$   
 $= 127.2345 \geq 127.1186$  kips (OK)

Bolt Bearing on Plate:

Bearing Strength / Bolt / Thickness Using Bolt Edge Distance = Fbe  
Edge Dist. = 1.25 in., Hole Size = 0.8125 in.  
 $= (1 / 2.0) * 1.2 * L_c * F_u \leq (1 / 2.0) * 2.4 * d * F_u = 52.2$  kips/in.  
 $= (1 / 2.0) * 1.2 * 0.8437 * 58 = 29.3625$  kips/in.

Bearing Strength / Bolt / Thickness Using Bolt Spacing = Fbs  
Bolt Spacing = 3 in., Hole Size = 0.8125 in.  
 $= (1 / 2.0) * 1.2 * L_c * F_u \leq (1 / 2.0) * 2.4 * d * F_u = 52.2$  kips/in.  
 $= (1 / 2.0) * 1.2 * 2.1875 * 58 = 76.125$  kips/in.  
Use: Fbs = 52.2 kips/in.

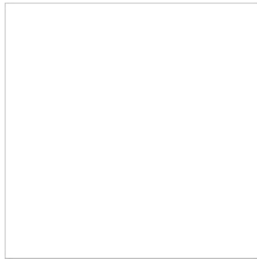
Allowable Bearing Strength at Bolt Holes =  $n N * n_L * (F_{be} + F_{bs} * (n_B - 1)) * t * e_f$   
 $= 1 * 2 * (29.3625 + 52.2 * (6 - 1)) * 0.875 * 1$   
 $= 508.1343 \geq 127.1186$  kips (OK)

Bolt Bearing on Flange:

Bearing Strength / Bolt / Thickness Using Bolt Edge Distance = Fbe  
Edge Dist. = 2 in., Hole Size = 0.8125 in.  
 $= (1 / 2.0) * 1.2 * L_c * F_u \leq (1 / 2.0) * 2.4 * d * F_u = 58.5$  kips/in.  
 $= (1 / 2.0) * 1.2 * 1.5937 * 65 = 62.1562$  kips/in.  
Use: Fbe = 58.5 kips/in.

Bearing Strength / Bolt / Thickness Using Bolt Spacing = Fbs  
Bolt Spacing = 3 in., Hole Size = 0.8125 in.  
 $= (1 / 2.0) * 1.2 * L_c * F_u \leq (1 / 2.0) * 2.4 * d * F_u = 58.5$  kips/in.  
 $= (1 / 2.0) * 1.2 * 2.1875 * 65 = 85.3125$  kips/in.  
Use: Fbs = 58.5 kips/in.

Allowable Bearing Strength at Bolt Holes =  $n N * n_L * (F_{be} + F_{bs} * (n_B - 1)) * t * e_f$   
 $= 1 * 2 * (58.5 + 58.5 * (6 - 1)) * 0.505 * 1$   
 $= 354.51 \geq 127.1186$  kips (OK)



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**1.e. Plate Tension Allowable Strength:**

**1.e.1. Tension Yielding:**  
 $(1 / FS) R_n = (1 / 1.67) * F_y * b * t$   
 $= (1 / 1.67) * 36 * 7.25 * 0.875$   
 $= 136.7514 \geq 127.1186 \text{ kips (OK)}$

**1.e.2. Tension Rupture:**  
 Effective Net Width:  
 $bn1 = b - \text{Max}(0.15 * b; nT * (dh + 0.0625))$   
 $= 7.25 - \text{Max}(0.15 * 7.25; 2 * (0.8125 + 0.0625)) = 5.5 \text{ in.}$   
 $bn2 = 2 * 0.85 * W_s = 2 * 0.85 * 3.625 = 6.1625 \text{ in.}$   
 $bn = \text{Min}(bn1, bn2) = \text{Min}(5.5, 6.1625) = 5.5 \text{ in.}$   
 $(1 / FS) R_n = (1 / 2.0) * F_u * bn * t$   
 $= (1 / 2.0) * 58 * 5.5 * 0.875$   
 $= 139.5625 \geq 127.1186 \text{ kips (OK)}$

**1.e.3. Block shear rupture of the Plate:**  
 $Agt = \text{Min}(g, 2 * e) * t = 3.5 * 0.875$   
 $= 3.0625 \text{ in}^2$   
 $Ant = Agt - (dh + 0.0625) * t$   
 $= 3.0625 - (0.875) * 0.875$   
 $= 2.2968 \text{ in}^2$   
 $Agv = 2 * ((nl - 1) * s + Le) * t$   
 $= 2 * ((6 - 1) * 3 + 1.25) * 0.875$   
 $= 28.4375 \text{ in}^2$   
 $Anv = Agv - 2 * (nl - 0.5) * (dh + 0.0625) * t$   
 $= 28.4375 - 2 * (6 - 0.5) * (0.875) * 0.875$   
 $= 20.0156 \text{ in}^2$   
 $(1 / FS) R_n = (1 / 2.0) * \text{Min}((0.6 * F_u * Anv + Ubs * F_u * Ant); (0.6 * F_y * Agv + Ubs * F_u * Ant))$   
 $= (1 / 2.0) * \text{Min}((0.6 * 58 * 20.0156 + 1 * 58 * 2.2968); (0.6 * 36 * 28.4375 + 1 * 58 * 2.2968))$   
 $= 373.7343 \geq 127.1186 \text{ kips (OK)}$

**1.e.4. Block shear rupture of the Beam Flange:**  
 $Agt = (bf - g) * t = (7.01 - 3.5) * 0.505$   
 $= 1.7725 \text{ in}^2$   
 $Ant = Agt - (nt - 1) * (dh + 0.0625) * t$   
 $= 1.7725 - (2 - 1) * (0.875) * 0.505$   
 $= 1.3306 \text{ in}^2$   
 $Agv = 2 * ((nl - 1) * s + ef) * t$   
 $= 2 * ((6 - 1) * 3 + 2) * 0.505$   
 $= 17.17 \text{ in}^2$   
 $Anv = Agv - 2 * (nl - 0.5) * (dh + 0.0625) * t$   
 $= 17.17 - 2 * (6 - 0.5) * (0.875) * 0.505$   
 $= 12.3093 \text{ in}^2$   
 $(1 / FS) R_n = (1 / 2.0) * \text{Min}((0.6 * F_u * Anv + Ubs * F_u * Ant); (0.6 * F_y * Agv + Ubs * F_u * Ant))$   
 $= (1 / 2.0) * \text{Min}((0.6 * 65 * 12.3093 + 1 * 65 * 1.3306); (0.6 * 50 * 17.17 + 1 * 65 * 1.3306))$

$= 283.2797 \geq 127.1186 \text{ kips (OK)}$

**1.f. Bottom Plate Allowable Compressive Strength:**  
 Unbraced Length (L) = c + ef = 0.5 + 2 = 2.5 in.  
 Effective Length Factor, K = 0.65  
 $KL / r = k * L / (t / 3.464) = 0.65 * 2.5 / (0.875 / 3.464) = 6.4333$   
 $KL / r \leq 25$   
 $F_{cr} = F_y = 36 \text{ ksi}$   
 $(1 / FS)cP_n = (1 / 1.67) * F_{cr} * A_g = (1 / 1.67) * 36 * 7.25 * 0.875 = 136.7514 \geq 127.1186 \text{ kips (OK)}$

**1.g. Top Plate-to-Support Weld:**  
 Required Fillet Weld Size =  $F_f / ((1 / 2.0) * 1.5 * 0.4242 * F_{exx} * b * 2)$   
 $= 127.1186 / ((1 / 2.0) * 1.5 * 0.4242 * 70 * 7.25 * 2)$   
 $= 0.3936 \text{ in.} \leq 0.4375 \text{ in. (OK)}$

**1.h. Bottom Plate-to-Support Weld:**  
 Required Fillet Weld Size =  $F_f / ((1 / 2.0) * 1.5 * 0.4242 * F_{exx} * b * 2)$   
 $= 127.1186 / ((1 / 2.0) * 1.5 * 0.4242 * 70 * 7.25 * 2)$   
 $= 0.3936 \text{ in.} \leq 0.4375 \text{ in. (OK)}$

Note: Descon does not check the moment versus rotation behavior of the connection.  
 If your particular application requires this check, you must do it outside the program.

**2. RIGHT SIDE BEAM - W24X55 SHEAR CONNECTION**

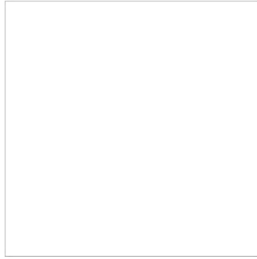
**2.a. Shear Connection Using One Plate:**  
 Plate (W x L x T): 17.5 in. X 3.5 in. X 0.3125 in.

Shear Connection Using One Plate:  
 Max. Thickness =  $db / 2 + 0.0625$   
 $= 0.4375 \geq \text{Min}(0.3125, 0.395) \text{ in. (OK)}$

Plate Material: A36  
 Beam Setback: 0.5 in.  
 Bolts: (6) (0.75 - A325 - N - STD)  
 Bolt Holes on Beam Web: 0.8125 in. Vert. X 0.8125 in. Horiz.  
 Bolt Holes on Plate: 0.8125 in. Vert. X 0.8125 in. Horiz.  
 Weld: 0.1875 E70XX - Fillet Welds

**Loading:**  
 Vertical Shear (V) = 60 kips  
 Axial Load (H) = 0 kips  
 Resultant (R) =  $(V^2 + H^2)^{0.5}$   
 $= (60^2 + 0^2)^{0.5}$   
 $= 60 \text{ kips}$   
 Theta =  $\text{Atan}(V / H) = \text{Atan}(60 / 0) = 90 \text{ degrees}$

**Check Bolt Spacing and Edge Distance:**



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**Spacing (s) = 3 ≥ Minimum Spacing = 2 in. (OK)**

Distance to Horiz. Edge of PL (ev):  
 (If Sheared Edge)  
 = **1.25 ≥ 1.25 in. (OK)**  
 (If Rolled or Gas Cut Edge)  
 = **1.25 ≥ 1 in. (OK)**

Minimum Distance to Vert. Edge of PL:  
 = Max(2 \* db, 1.25) = 1.5 in.  
 Distance to Vert. Edge of PL (eh):  
 = **1.5 ≥ 1.5 in. (OK)**

Minimum Distance to End of Beam:  
 = Max(2 \* db, 1.25) = 1.5 in.  
 Distance to End of Beam (Lh):  
 = **1.5 ≥ 1.5 in. (OK)**

**Connection Top Location: (OK)**  
**Connection Bottom Location: (OK)**  
**Connection Depth = 17.5 ≥ T / 2 (OK)**

**2.b. Bolt Strength:**  
 Load Eccentricity for Bolts (eb):  
 Bolt Rows = 6 ≤ 9 ... eb = 0

**2.c. Allowable Bolt Shear Strength - Single Plate**  
 (1 / FS)Rn = nIN \* nSP \* nB \* Fv  
 = 1 \* 1 \* 6 \* 10.6028  
 = **63.6172 ≥ 60 kips (OK)**

**2.d. Allowable Shear Strength of the Beam:**

**2.d.1. Allowable Shear Yield Strength:**  
 A = dw \* tw = 23.6 \* 0.395 = 9.322 in<sup>2</sup>  
 Rn = 0.6 \* Fy \* A \* Cv  
 = 0.6 \* 50 \* 9.322 \* 1  
 = 279.66 kips  
 (1 / FS) Rn = (1 / 1.67) \* 279.66 = 167.461 kips  
 = **167.461 ≥ 60 kips (OK)**

**2.d.2. Allowable Shear Rupture Strength:**  
 Anv = (dw - N \* (dh + 0.0625)) \* tw  
 = (23.6 - 6 \* (0.8125 + 0.0625)) \* 0.395  
 = 7.2482 in<sup>2</sup>  
 Rn = 0.6 \* Fu \* Anv  
 = 0.6 \* 65 \* 7.2482  
 = 282.6817 kips  
 (1 / FS) Rn = (1 / 2.0) \* 282.6817 = 141.3408 kips  
 = **141.3408 ≥ 60 kips (OK)**

**2.d.3. Allowable Shear Strength of the Plate:**

**2.d.4. Allowable Shear Yield Strength:**  
 A = dw \* tw = 17.5 \* 0.3125 = 5.4687 in<sup>2</sup>  
 Rn = 0.6 \* Fy \* A \* Cv  
 = 0.6 \* 36 \* 5.4687 \* 1  
 = 118.125 kips  
 (1 / FS) Rn = (1 / 1.5) \* 118.125 = 78.75 kips  
 (1 / FS)Vn = **78.75 ≥ 60 kips (OK)**

**2.d.5. Allowable Shear Rupture Strength:**  
 Net Area (An) = (L - nL \* (dh + 0.0625)) \* t  
 = (17.5 - 6 \* 0.875) \* 0.3125 = 3.8281 in<sup>2</sup>  
 Shear Rupture Strength = Npl \* An \* (1 / 2.0) \* 0.6 \* Fu = 1 \* 3.8281 \* (1 / 2.0) \* 0.6 \* 58  
 = **66.6093 ≥ 60 kips (OK)**

**2.d.6. Block Shear Strength of the Plate:**  
 Gross Area with Tension Resistance (Agt)  
 = (et + (Nh - 1) \* sh) \* t  
 = (1.5 + (1 - 1) \* 3) \* 0.3125  
 = 0.4687 in<sup>2</sup>

Net Area with Tension Resistance (Ant)  
 = Agt - (Nh - 0.5) \* (dh + 0.0625) \* t  
 = 0.4687 - (1 - 0.5) \* (0.8125 + 0.0625) \* 0.3125  
 = 0.332 in<sup>2</sup>

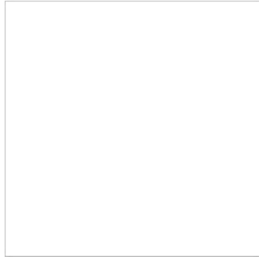
Gross Area with Shear Resistance (Agv)  
 = (L - el) \* t = (17.5 - 1.25) \* 0.3125 = 5.0781 in<sup>2</sup>

Net Area with Shear Resistance (Anv)  
 = Agv - (Nv - 0.5) \* (dv + 0.0625) \* t  
 = 5.0781 - (6 - 0.5) \* (0.8125 + 0.0625) \* 0.3125  
 = 3.5742 in<sup>2</sup>

(1 / FS) Rn = (1 / 2.0) \* Min((0.6 \* Fu \* Anv + Ubs \* Fu \* Ant); (0.6 \* Fy \* Agv + Ubs \* Fu \* Ant))  
 = (1 / 2.0) \* Min((0.6 \* 58 \* 3.5742 + 1 \* 58 \* 0.332); (0.6 \* 36 \* 5.0781 + 1 \* 58 \* 0.332))  
 = **64.4726 ≥ 60 kips (OK)**

Bolt Bearing on Plate:  
 Bearing Strength / Bolt / Thickness Using Bolt Edge Distance = Fbe  
 Edge Dist. = 1.25 in., Hole Size = 0.8125 in.  
 = (1 / 2.0) \* 1.5 \* Lc \* Fu ≤ (1 / 2.0) \* 3.0 \* d \* Fu = 65.25 kips/in.  
 = (1 / 2.0) \* 1.5 \* 0.8437 \* 58 = 36.7031 kips/in.

Bearing Strength / Bolt / Thickness Using Bolt Spacing = Fbs  
 Bolt Spacing = 3 in., Hole Size = 0.8125 in.  
 = (1 / 2.0) \* 1.5 \* Lc \* Fu ≤ (1 / 2.0) \* 3.0 \* d \* Fu = 65.25 kips/in.  
 = (1 / 2.0) \* 1.5 \* 2.1875 \* 58 = 95.1562 kips/in.  
 Use: Fbs = 65.25 kips/in.



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Allowable Bearing Strength at Bolt Holes =  $nIN * nL * (Fbe + Fbs * (nB - 1)) * t * ef$   
 $= 1 * 1 * (36.7031 + 65.25 * (6 - 1)) * 0.3125 * 1$   
 $= 113.4228 \geq 60 \text{ kips (OK)}$

**2.e. Bolt Bearing on Beam Web:**

Bearing Strength / Bolt / Thickness Using Bolt Spacing = Fbs  
 Bolt Spacing = 3 in., Hole Size = 0.8125 in.  
 $= (1 / 2.0) * 1.5 * Lc * Fu \leq (1 / 2.0) * 3.0 * d * Fu = 73.125 \text{ kips/in.}$   
 $= (1 / 2.0) * 1.5 * 2.1875 * 65 = 106.6406 \text{ kips/in.}$   
 Use: Fbs = 73.125 kips/in.

Allowable Bearing Strength at Bolt Holes =  $nIN * nL * Fbs * nB * t * ef$   
 $= 1 * 1 * 73.125 * 6 * 0.395 * 1$   
 $= 173.3062 \geq 60 \text{ kips (OK)}$

**2.e.1. Weld Strength:**

**Weld Size (w) = 0.1875  $\geq$  Minimum Weld, 0.1875 in. (OK)**

Eccentric Weld

k = 0

a = 0

Theta = 0

(1 / FS) C = 0.9266

Maximum useful weld size for support thickness:

$= Fu * t_{eff} / (0.707 * Fexx)$

$= 65 * 0.71 / (0.707 * 70)$

$= 0.9325 \geq 0.1875 \text{ in. (OK)}$

Maximum useful weld size for plate thickness:

$= Fu * tp / (2 * 0.707 * Fexx)$

$= 58 * 0.3125 / (2 * 0.707 * 70)$

= 0.1831 in.

0.1831 << 0.1875 in.

Use 0.1831 in. for strength calculation.

(1 / FS) Rn =  $2 * C * C1 * D * L = 2 * 0.9266 * 1 * 2.9298 * 17.5$   
 $= 95.0259 \geq 60 \text{ kips (OK)}$

**3. COLUMN WEB SHEAR REINFORCEMENT**

Framing System: OMF

Column Axial Force (Pu) = 0 kips

Column Shear Force (Vus) = 0 kips

**3.a. Right Side Beam Flange Forces:**

PufRight =  $Mu / dm + Pu / 2$

=  $3000 / 24.475 + 0 / 2$

= 122.574 kips

**3.b. Column Panel Zone:**

Required Strength (Vu)

=  $|PufLeft + PufRight - Vus|$   
 $= |0 + 122.574 - 0|$   
 $= 122.574 \text{ kips}$

**3.b.1. Column Web Shear Strength:**

Pc =  $0.6 * Py = 0.6 * A * Fy = 0.6 * 26.5 * 50 = 795 \text{ kips}$

Pr  $\leq 0.4 * Pc$

(1 / FS)Rv =  $(1 / 1.67) * 0.6 * Fy * d * tw$

=  $(1 / 1.67) * 0.6 * 50 * 14 * 0.44$

= 110.6586 << 122.574 kips

Doubler Plate Required for Strength

**3.b.2. Shear Buckling of Web:**

Thickness Required =  $h * (Fy^{0.5}) / (2.24 * E^{0.5}) = 11.38 * (50^{0.5}) / (2.24 * 29000^{0.5})$

= 0.2109  $\leq 0.44 \text{ in.}$

Doubler Plate Not Required for Shear Buckling

Doubler Plate Thickness = 1 in.

Required for strength =  $Vudp / (Npl * (1 / 1.67) * 0.6 * Fy * dc)$

=  $11.9153 / (1 * (1 / 1.67) * 0.6 * 36 * 14)$

= **0.0658  $\leq 1 \text{ in. (OK)}$**

Required for fillet weld detail =  $k - tf - re$

= **2 - 0.71 - 0.375 = 0.915  $\leq 1 \text{ in. (OK)}$**

Required for shear buckling =  $(d - 2 * k) * (Fy)^{0.5} / (2.24 * E^{0.5})$

=  $(14 - 2 * 1.31) * (36)^{0.5} / (2.24 * 29000^{0.5})$

= **0.1789  $\leq 1 \text{ in. (OK)}$**

Required to transmit stiffener force:

=  $(Rust1 + Rust2) / ((1 / 1.67) * 0.6 * Fy * 2 * \text{Min}(2 * (L - 2 * \text{clip}), dc))$

=  $(28.2439 + 0) / ((1 / 1.67) * 0.6 * 36 * 2 * \text{Min}(2 * (12.5625 - 2 * 1.29); 14))$

= **0.0779  $\leq 1 \text{ in. (OK)}$**

Doubler Plate Thickness excluding Fillet Weld Detail:

= 0.1875 in.

**4. COLUMN STIFFENERS**

Framing System: OMF

Column Axial Force (Pu) = 0 kips

Column Shear Force (Vus) = 0 kips

**4.a. Right Side Beam Flange Forces:**

PufRight =  $Mu / dm + Pu / 2$

=  $3000 / 24.475 + 0 / 2$

= 122.574 kips

**4.b. Column Stiffeners**

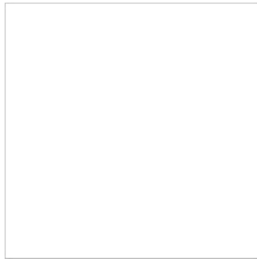
Right Side Beam

Local Flange Bending Strength, (1 / FS) Rn

=  $(1 / 1.67) * 6.25 * (tf^2) * Fy * ct$

=  $(1 / 1.67) * 6.25 * (0.71^2) * 50 * 1$

= 94.330 kips



**PROJECT NAME**  
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**CHECKED BY**  
**DESCRIPTION**

Sample Configuration  
  
1/17/2019

**PAGES**  
**CODE**  
**METHOD**  
**UNITS**  
**SEISMIC**  
**FILE NAME**

8 / 9  
AISC13  
ASD  
US  
No  
Sample Moment  
Configuration.dsn

Local Web Yielding Strength,  $(1 / FS) R_n$   
 $= (1 / 1.5) * (c_t * 5 * k + t + 2 * w) * t_w * F_y$   
 $= (1 / 1.5) * (1 * 5 * 1.31 + 0.875 + 2 * 0.4375) * 0.44 * 50$   
 $= 121.7333 \text{ kips}$

**4.b.1. Column Web Crippling:**

$N = t + 2 * w = 0.875 + 2 * 0.4375 = 1.75 \text{ in.}$   
 $C_t = 1.0$   
 $N_d = 3 * N / d = 3 * 1.75 / 14 = 0.375$   
 $(1 / FS) R_n = (1 / 2.0) * 0.8 * c_t * (t_w^2) * [1 + N_d * (t_w / t_f)^{1.5}] * (E * F_y * t_f / t_w)^{0.5}$   
 $= (1 / 2.0) * 0.8 * 1 * (0.44^2) * [1 + 0.375 * (0.44 / 0.71)^{1.5}] * (29000 * 50 * 0.71 / 0.44)^{0.5}$   
 $= 140.1254 \text{ kips}$

Tension Flange Stiffener Force (TFrc):

Right Side:  
 $RTFrc = \text{Max}(RPuf - R(1 / FS) R_n \text{FlBending}; RPuf - R(1 / FS) R_n \text{WebYielding}) \geq 0$   
 $= \text{Max}(122.574 - 94.330; 122.574 - 121.7333) = 28.2439 \text{ kips}$

Compression Flange Stiffener Force (CFrc):

Right Side:  
 $RCFrc = \text{Max}[(RPuf - R(1 / FS) R_n \text{WebCrippling}), (RPuf - R(1 / FS) R_n \text{WebYielding})] \geq 0$   
 $= \text{Max}[(122.574 - 140.1254), (122.574 - 121.7333)] = 0.8407 \text{ kips}$

$TFrc = \text{Max}(LTFrc, RTFrc) = \text{Max}(0, 28.2439) = 28.2439 \text{ kips}$

$CFrc = \text{Max}(LCFrc, RCFrc) = \text{Max}(0, 0.8407) = 0.8407 \text{ kips}$

$TFrc >> 0$  or High Seismic Loading

Stiffeners required opposite tension flange

$CFrc >> 0$  or High Seismic Loading

Stiffeners required opposite compression flange.

Required stiffener area for strength:

Tension and/or compression:

$A_{st} = \text{Max}(TFrc, CFrc) / ((1 / 1.67) * F_y)$   
 $= \text{Max}(28.2439, 0.8407) / ((1 / 1.67) * 36)$   
 $= \text{Max}(1.3102, 0.039) \text{ in}^2$

**Stiffener Width (bs) = 6.25 ≥ Minimum Width = 2.1966 in. (OK)**

Stiffener Length:

$L = d - 2 * t_f = 14 - 2 * 0.71 = 12.5625 \text{ in.}$

(Using Full Length Stiffeners)

Stiffener thickness required for shear:

$= \text{Max}([LTFrc + RCFrc], [LCFrc + RTFrc]) / ((1 / 1.67) * 0.6 * F_y * (L - 2 * clip) * 2)$   
 $= \text{Max}([0 + 0.8407]; [0 + 28.2439]) / ((1 / 1.67) * 0.6 * 36 * (12.5625 - 2 * 1.29) * 2)$   
 $= 0.1093 \leq 0.4375 \text{ in. (OK)}$

Stiffener thickness required for minimum area:

$= A_{st} / (2 * (bs - clip))$   
 $= 1.3102 / (2 * (6.25 - 1.29)) = 0.132 \leq 0.4375 \text{ in. (OK)}$

Minimum Thickness =  $\text{Max}(t_m / 2, bs * (F_y / E)^{0.5} / 0.56)$   
 $= \text{Max}(0.875 / 2, 6.25 * (36 / 29000)^{0.5} / 0.56)$   
 $= 0.4375 \leq 0.4375 \text{ in. (OK)}$

**5. DOUBLER PLATE WELDS:**

Doubler Plate to Flange Weld = 0.25 in.

**Minimum weld size for material thickness = 0.25 ≤ 0.25 in. (OK)**

Weld size to develop doubler plate force:

$= \text{Max}(V_{upl} / ((1 / 2.0) * 0.707 * 0.6 * F_{exx} * \text{Column } d), t_{eff} * 1.4142)$   
 $= \text{Max}(11.9153 / ((1 / 2.0) * 0.707 * 0.6 * 70 * 14), 0.0658 * 1.4142)$   
 $= \text{Max}(0.0573, 0.093) \text{ in.}$   
 $= 0.0625 \leq 0.25 \text{ in. (OK)}$

Doubler Plate to Web Weld:

Minimum weld size =  $\text{Min}(W_{min}, t - 0.0625)$   
 $= 0.1875 \leq 0.1875 \text{ in. (OK)}$

**6. STIFFENER WELDS:**

**6.a. Stiffener to Flange Weld:**

**Minimum Weld Size = 0.1875 ≤ 0.1875 in. (OK)**

**6.b. Tension Stiffener to Flange Weld:**

$w_{Req.} = \text{Rust} / ((1 / FS) * (1.5 * 0.6 * F_{exx}) * bs - clip) * 2 * 1.4142$   
 $= 28.2439 / ((1 / 2.0) * 1.5 * 0.6 * 70) * (6.25 - 1.29) * 2 * 1.4142$   
 $= 0.0639 \leq 0.1875 \text{ in. (OK)}$

**6.c. Compression Stiffener to Flange Weld:**

$w_{Req.} = \text{Rust} / ((1 / FS) * (1.5 * 0.6 * F_{exx}) * bs - clip) * 2 * 1.4142$   
 $= 28.2439 / ((1 / 2.0) * 1.5 * 0.6 * 70) * (6.25 - 1.29) * 2 * 1.4142$   
 $= 0.0019 \leq 0.1875 \text{ in. (OK)}$

**6.d. Stiffener to Panel Zone Weld:**

Stiffener Force (Rust) =  $\text{Max}([LTRust + RcRust]; (RTRust + LCRust))$   
 $= \text{Max}([0 + 0.8407]; (28.2439 + 0)) = 28.2439 \text{ kips}$

**6.e. Welds need to develop only the lesser of Rust and the minimum of the following forces:**

Allowable Strength of stiffeners to flange connection:

$= (1 / 1.67) * F_y * 2 * (L - clip) * t$   
 $= (1 / 1.67) * 36 * 2 * (12.5625 - 1.29) * 0.4375 = 212.625 \text{ kips}$

Allowable Shear Strength of stiffener and web interface area:

$= (1 / 1.67) * 0.6 * F_y * (bs - 2 * clip) * 2 * t$   
 $= (1 / 1.67) * 0.6 * 36 * (6.25 - 2 * 1.29) * 2 * 0.4375 = 112.9755 \text{ kips}$

Shear yield strength of the panel zone:

$= (1 / 1.67) * 0.6 * F_{yc} * d_c * t_w$  (for column web, if applicable)  
 $= (1 / 1.67) * 0.6 * 50 * 14 * 0.44 = 110.6586 \text{ kips}$   
 $= (1 / 1.67) * 0.6 * F_{yp} * d_c * t_p$  (for doubler plate, if applicable)





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Sample Configuration

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9 / 9  
AISC13  
ASD  
US  
No  
Sample Moment  
Configuration.dsn

$$= (1 / 1.67) * 0.6 * 36 * 14 * 1 = 181.0778 \text{ kips}$$

Weld Design Force, Rust\_Weld = 28.2439 kips

**Minimum Weld Size = 0.1875 ≤ 0.1875 in. (OK)**

Required weld size for strength:

$$= \text{Rust\_Weld} / (0.8485 * F_{exx} * (L - 2 * \text{clip}))$$

$$= 28.2439 / (0.8485 * 70 * (12.5625 - 2 * 1.29))$$

$$= \mathbf{0.0476 \leq 0.1875 \text{ in. (OK)}}$$

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